

# Section 7

# Serviceability limit states

Heinrich Kreuzinger

# Serviceability limit states

Calculation of

- Deformations, Deflections
- Vibrations



# Deflections, Deformations

$$w = \int_{\text{System}} \left( \frac{M \cdot \bar{M}}{EI} + \frac{N \cdot \bar{N}}{EA} + \frac{Q \cdot \bar{Q}}{GA^*} \right) ds + \sum_{\text{Verbindungen}} \frac{F \cdot \bar{F}}{K_{\text{ser}}}$$

7.1 Table 1



## Vibrations

$$\omega = \sqrt{\frac{K}{M}} = 2\pi \cdot f$$

## Section 2.2.3

(5) For structures consisting of members, components and connections with the same creep behaviour and under the assumption of a linear relationship between the actions and the corresponding deformations, as a simplification of 2.2.3(3), the final deformation,  $u_{fin}$ , may be taken as:

$$u_{fin} = u_{fin,G} + u_{fin,Q_1} + u_{fin,Q_i} \quad (2.2)$$

where:

$$u_{fin,G} = u_{inst,G} (1 + k_{def}) \quad \text{for a permanent action, } G \quad (2.3)$$

$$u_{fin,Q_1} = u_{inst,Q_1} (1 + \psi_{2,1} k_{def}) \quad \text{for the leading variable action, } Q_1 \quad (2.4)$$

$$u_{fin,Q_i} = u_{inst,Q_i} (\psi_{0,i} + \psi_{2,i} k_{def}) \quad \text{for accompanying variable actions, } Q_i \text{ (} i > 1 \text{)} \quad (2.5)$$

$u_{inst,G}$ ,  $u_{inst,Q_1}$ ,  $u_{inst,Q_i}$  are the instantaneous deformations for action  $G$ ,  $Q_1$ ,  $Q_i$  respectively;

$$E = \frac{E_{mean}}{(1 + k_{def})}$$

- $w_c$  is the precamber (if applied);
- $w_{inst}$  is the instantaneous deflection;
- $w_{creep}$  is the creep deflection;
- $w_{fin}$  is the final deflection;
- $w_{net,fin}$  is the net final deflection.

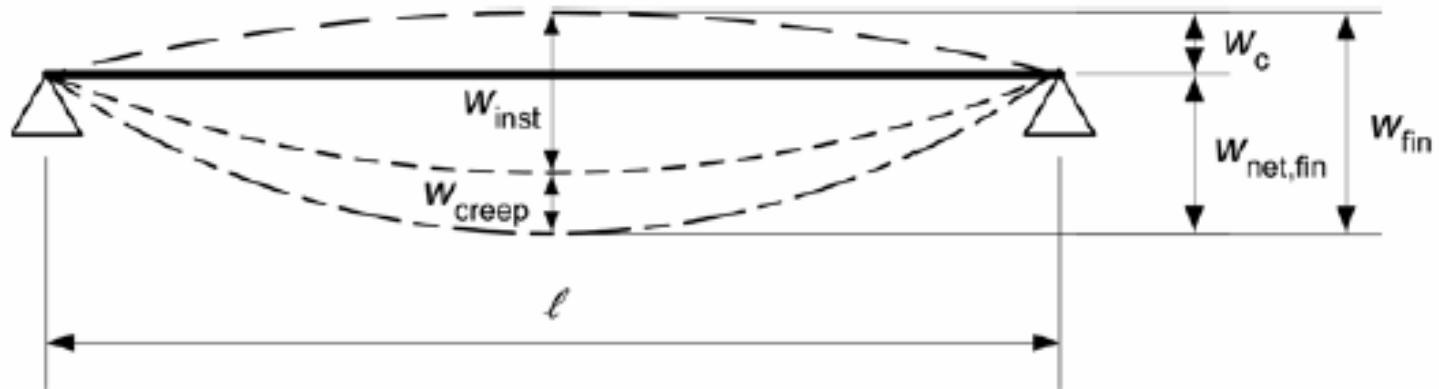


Figure 7.1 – Components of deflection

$$w_{net,fin} = w_{inst} + w_{creep} - w_c = w_{fin} - w_c$$

NOTE: The recommended range of limiting values of deflections for beams with span  $\ell$  is given in Table 7.2 depending upon the level of deformation deemed to be acceptable. Information on National choice may be found in the National annex.

**Table 7.2 – Examples of limiting values for deflections of beams**

	$w_{inst}$	$w_{net,fin}$	$w_{fin}$
Beam on two supports	$\ell/300$ to $\ell/500$	$\ell/250$ to $\ell/350$	$\ell/150$ to $\ell/300$
Cantilevering beams	$\ell/150$ to $\ell/250$	$\ell/125$ to $\ell/175$	$\ell/75$ to $\ell/150$

# Deflections

# Vibrations

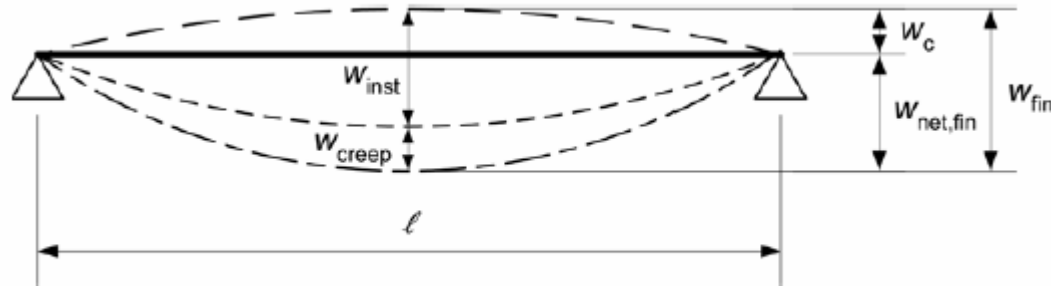


Figure 7.1 – Components of deflection

# Frequency

$$f = \frac{5}{\sqrt{0,8 \cdot w}}$$

# Vibrations:

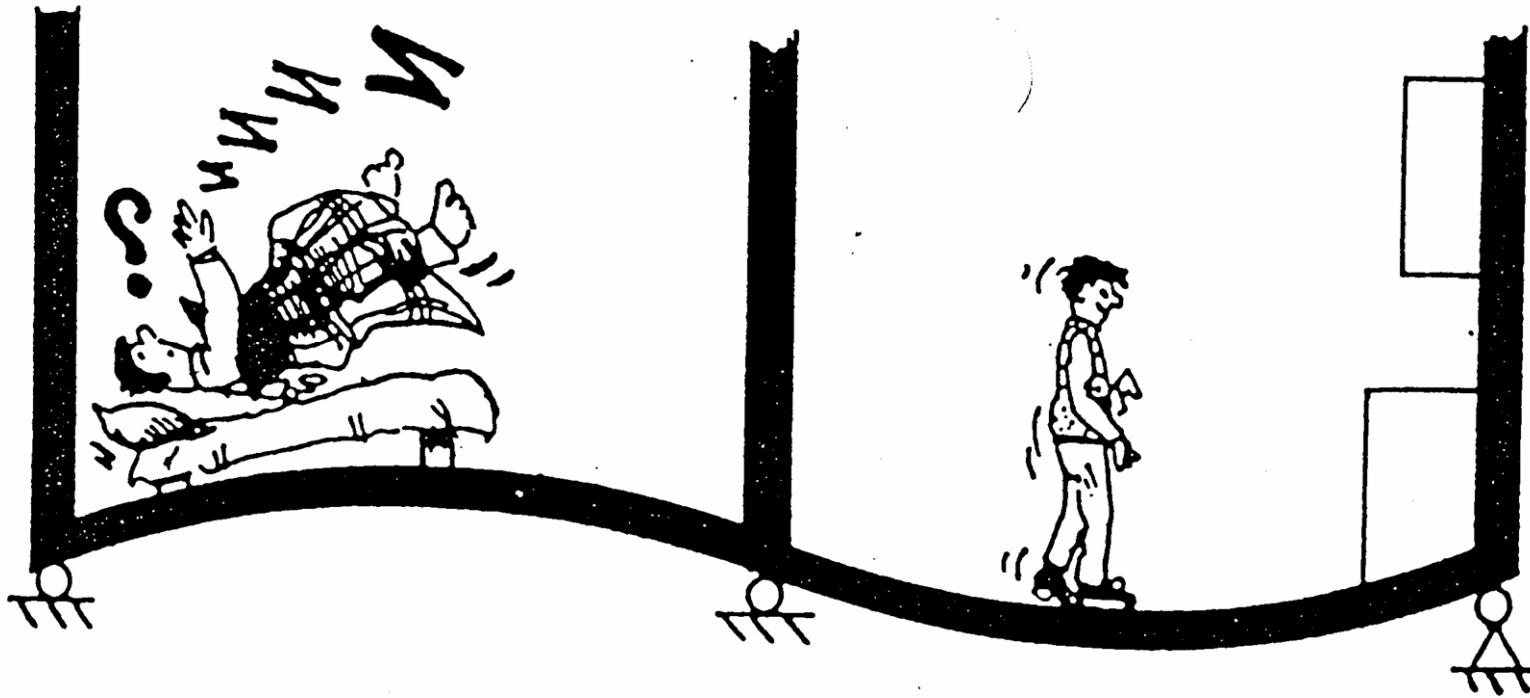
## Servicability limit states

If necessary

## Ultimate limit states

Fatigue  
Durability



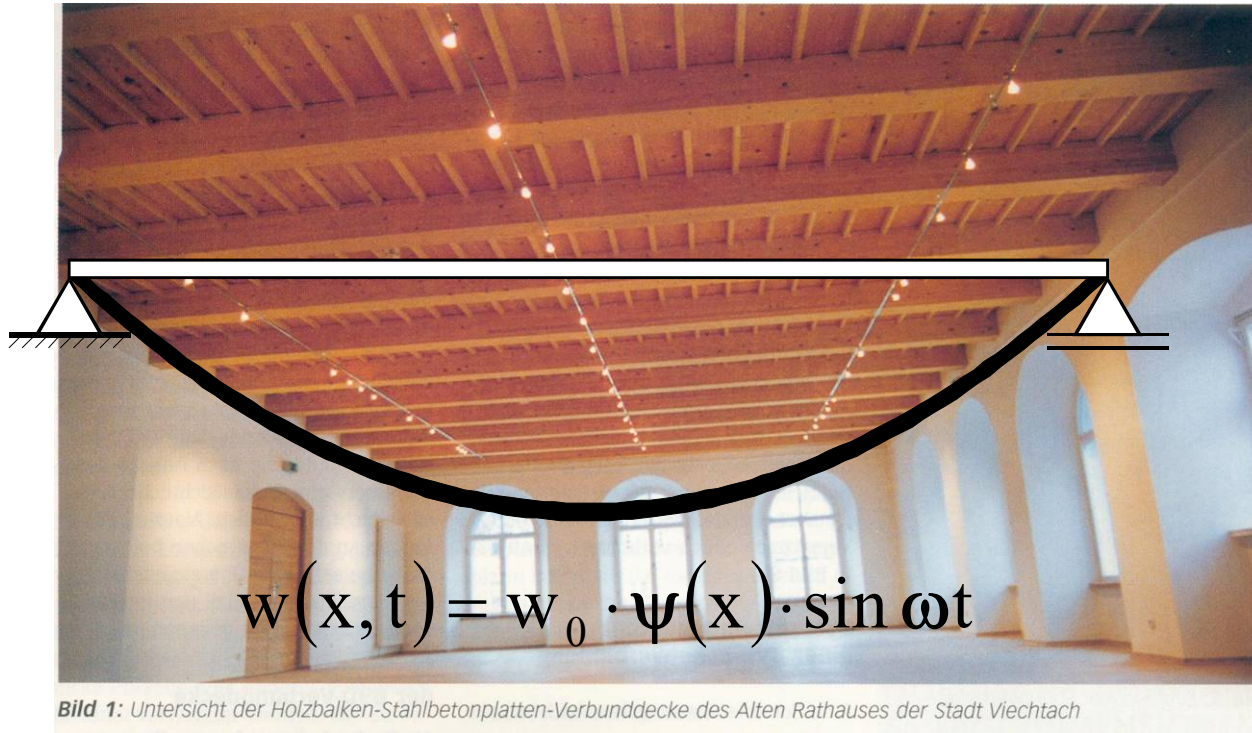


Serviceability

(Ohlsen)

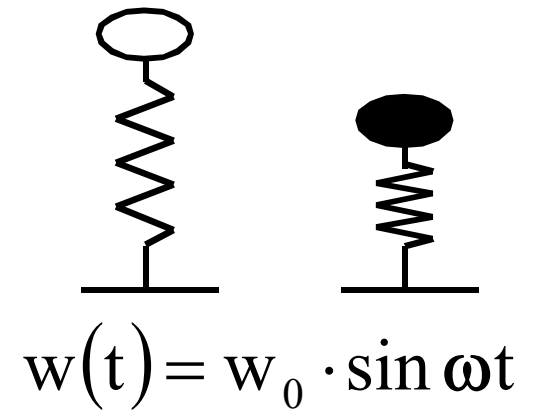
# Single degree of freedom System

## floor, beam



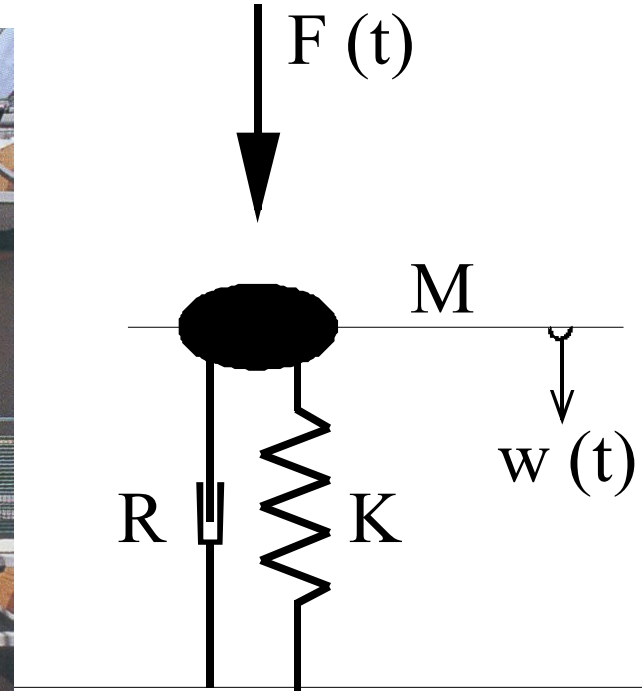
$$w(x, t) = w_0 \cdot \psi(x) \cdot \sin \omega t$$

Bild 1: Untersicht der Holzbalken-Stahlbetonplatten-Verbunddecke des Alten Rathauses der Stadt Viechtach



Viechtach, Bertsche

# Single degree of freedom System



Lemgo, Mayer/Ludscher, SFS

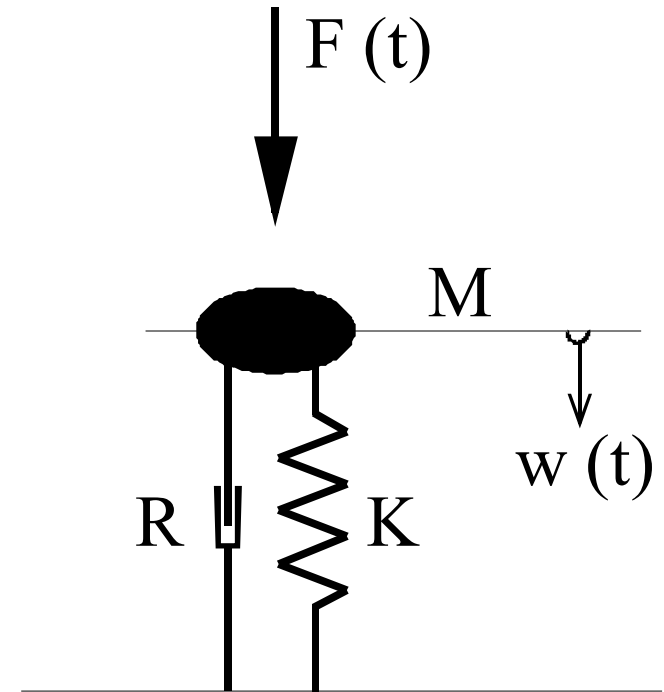
# Single degree of freedom System

- M Mass t  
 K Stiffness kN/m  
 R Damping kN/(m/s)  
 $R=2 M D \omega$   
 D Damping ratio

$$\omega = \sqrt{\frac{K}{M}} = 2 \cdot \pi \cdot f$$

$$w_g = \frac{G}{K} = \frac{M \cdot g}{K} = \frac{g}{\omega^2} = \frac{g}{(2 \cdot \pi \cdot f)^2}$$

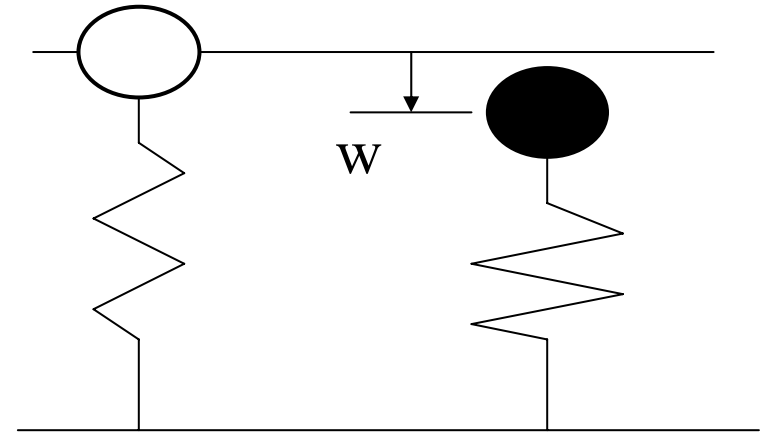
$$f = \frac{5}{\sqrt{w_g}} \quad w_g \text{ in cm!}$$



# Single degree of freedom System

Frequency - Deformation

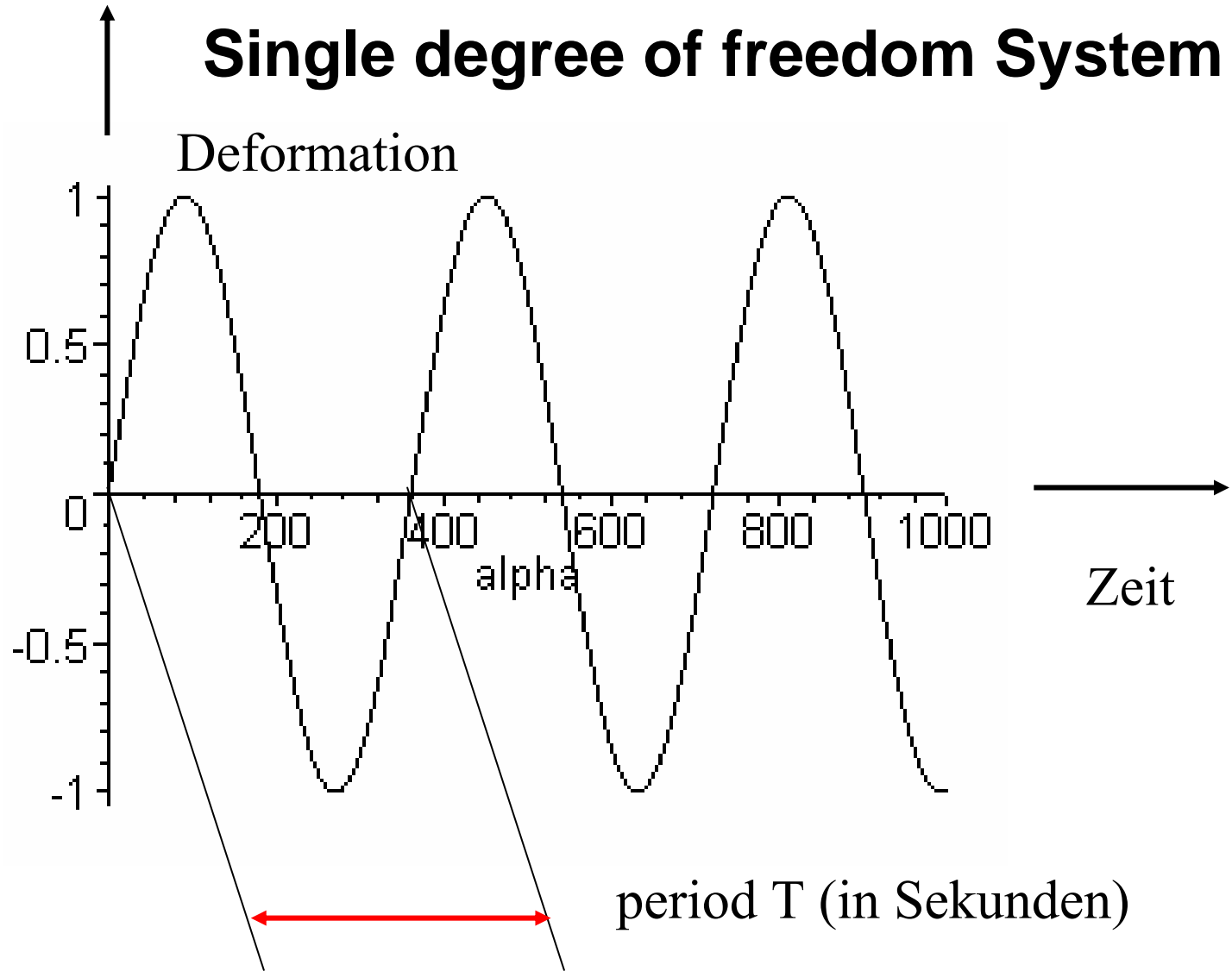
$$f = \frac{5}{\sqrt{w_g}} \quad w_g \text{ in cm!}$$



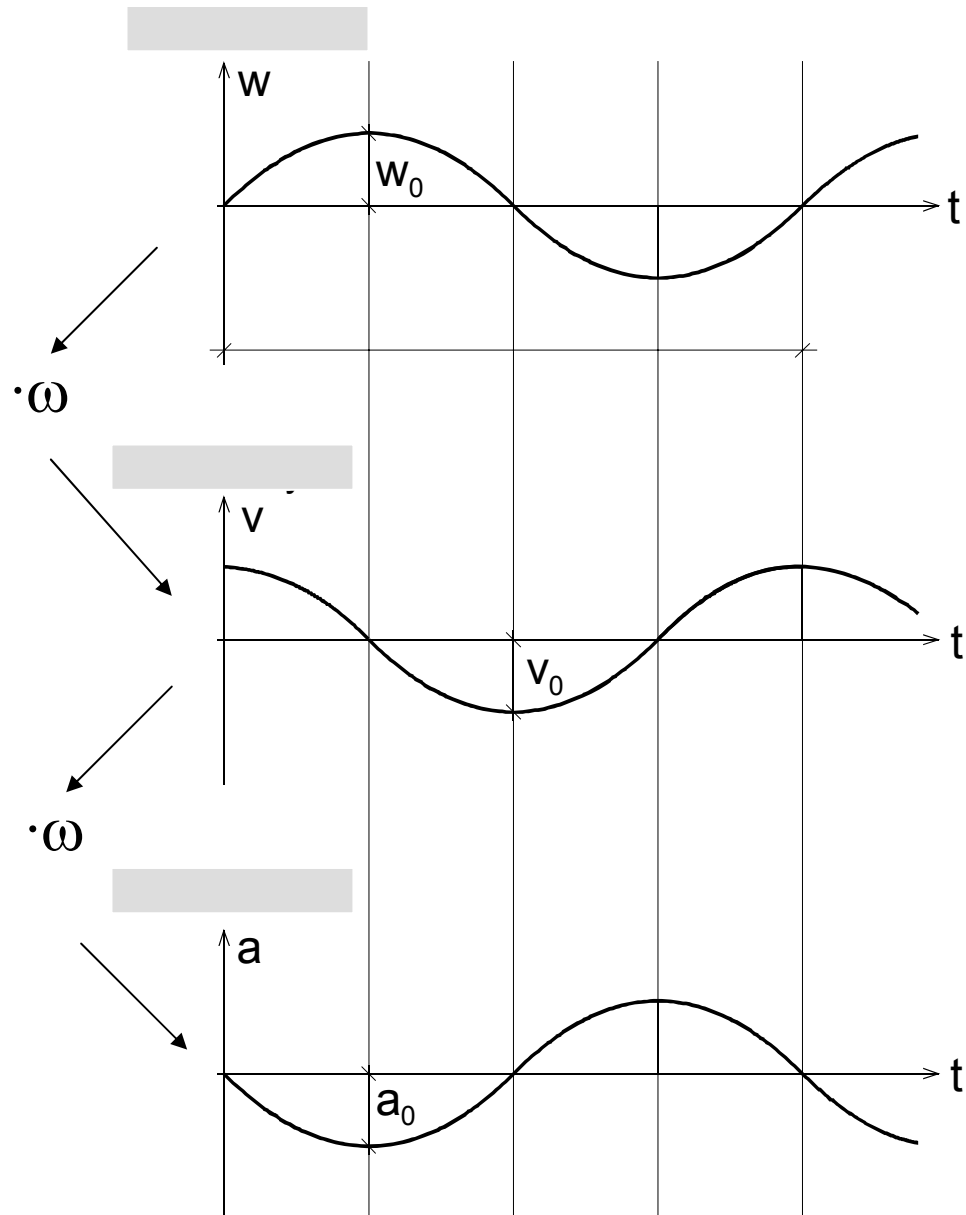
$$w_{g+0,3p} = 6 \text{ mm} = 0,6 \text{ cm}$$

$$f = \frac{5}{\sqrt{0,8 \cdot 0,6}} = 7,2 \text{ Hz}$$

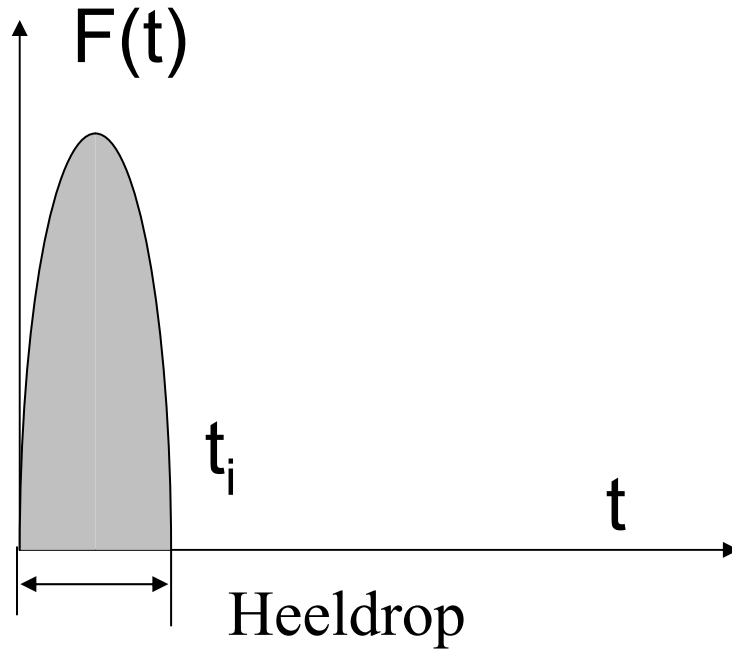
↑  
Factor for beam



Deflection  $w$   
 Velocity  $v$   
 Acceleration  $a$



# Impuls



Impuls I

$T_i=0$

$F=\text{infinite}$

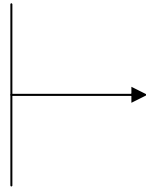
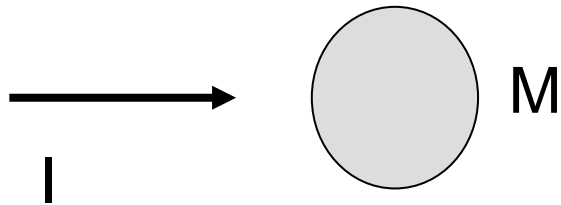
$$I = \int F(t) \cdot dt = M \cdot v = M \cdot \sqrt{2 \cdot g \cdot h} =$$

$$55 \cdot \sqrt{2 \cdot 9,81 \cdot 0,05} = 55 \text{ kg} \cdot \text{m/s} = 55 \text{ N} \cdot \text{s}$$



# Impuls

Velocity  $v$



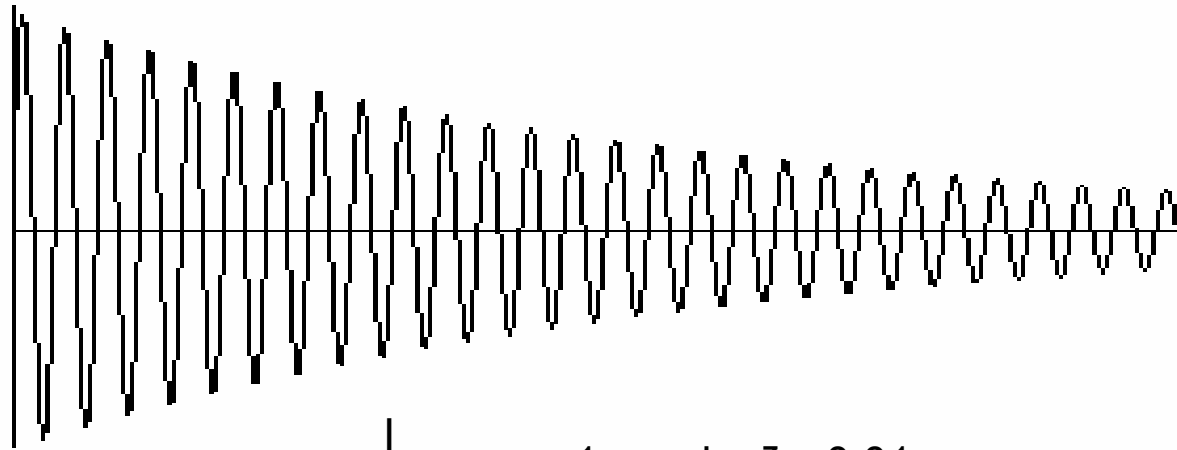
$$v = \frac{I}{M}$$

$$\frac{\text{N} \cdot \text{s}}{\text{kg}} = \frac{\text{m}}{\text{s}}$$

$t = 0$

# Impuls

$$w(t) = \frac{I}{M \cdot \omega \cdot \sqrt{1 - \zeta^2}} \cdot e^{-\zeta \cdot \omega \cdot t} \cdot \sin\left(\sqrt{1 - \zeta^2} \cdot \omega \cdot t\right)$$



$$\frac{I}{M \cdot \omega \cdot \sqrt{1 - \zeta^2}} = 1 \quad \text{und} \quad \zeta = 0,01$$

## Impuls – I = 1Ns, 7.3.3(5)

$$f_{1,n} = f_0 \cdot \sqrt{1 + \frac{EJ_b \ell^4}{EJ_\ell b^4} \cdot n_4^4}$$

Number of frequencies  $f_{1,n} \leq 40$  Hz

$$v = \frac{4 \cdot (0,4 + 0,6 \cdot n_{40})}{m \cdot b \cdot \ell + 200} \approx \frac{1}{m \cdot \frac{\ell}{2} \cdot b_{SI}} = \frac{1}{M}$$

$f_{1,n}$  frequency of a plate  
 $n$  number of waves in direction vertical to the main span

$$n_{40} = \left[ \left( \left( \frac{40}{f_0} \right)^2 - 1 \right) \cdot \frac{b^4 EJ_\ell}{\ell^4 EJ_b} \right]^{0,25}$$

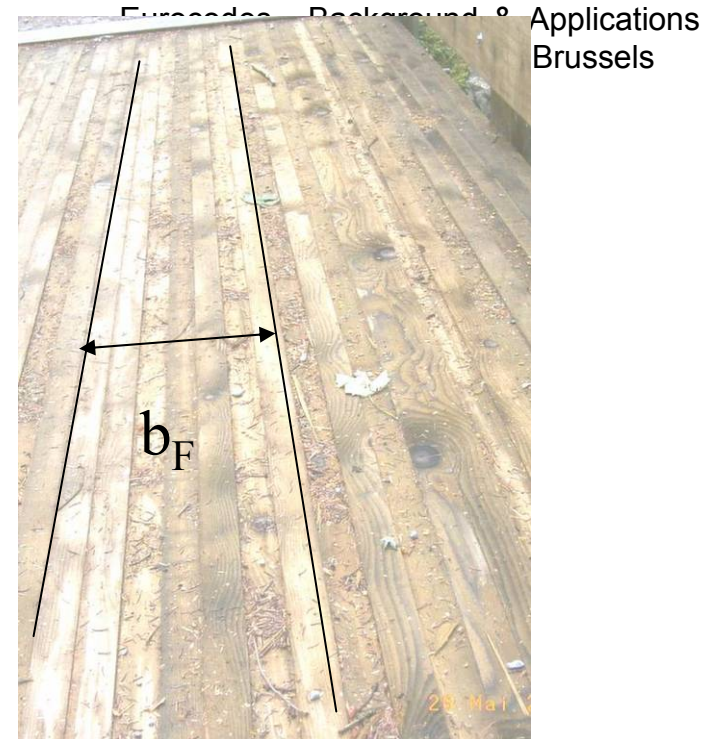
# Impuls – Heeldrop $I = 55 \text{ Ns}$

Experimental solution,  
Found by testing

$$v = \frac{0,6}{m^{0,5} \cdot EI_{\ell}^{0,25} \cdot EI_b^{0,25}} \frac{\text{m}}{\text{s}}$$

## 7.3.3(2)

$$w = \frac{F_0}{K}$$



uniform load

$$w = \frac{5 \cdot q \cdot l^4}{384 \cdot EI}$$

Single load

$$w_F = \frac{F \cdot l^3}{48 \cdot EI_1 \cdot b_F} \quad b_F = \frac{l}{1,1} \cdot \sqrt[4]{\frac{EI_b}{EI_l}}$$

# Serviceability limit states

	1	2	3	4
	Value	Limit	aim	EN 1995-1-1 2004 (E)
1	deflection $w$	$w < l/X$	Enough rigidity small deflection	7.2
4	frequency $w_{G,inst}$ quasi ständig, $g+\psi_2p$	$f > 8$ Hz ( $w < 0,5$ cm)  $f < 8$ Hz ( $w > 0,5$ cm)	No resonance	7.3.3(2)  7.3.3(1)

3	Deflection Single load F=1kN	$u < 0,5$ bis 4 mm	Small deformation Rigidity perpendicular to the main span	7.3.3(2) SIA 265
5	Impuls I=1 Ns (up to 40 Hz)	$v < b^{(f_1 \zeta - 1)}$  50 < b < 150 $\zeta = 0,01$	Velocity	7.3.3(2) SIA 265

# Information

2	Durchbiegung $w_{G,inst}$ quasi ständig, $g+\psi_2p$	$w < 6 \text{ mm}$  $w > 6 \text{ mm}$	Frequenz keine Resonanz- untersuchung Frequenz Resonanz- untersuchung	DIN 1052, 2004 9.3
6	velocity heeldrop $I=55Ns,$ $t_i = 0,05s$	$v < 6 b_1^{(f_1 \zeta - 1)}$	velocity	Mohr /bmh/



## 7.3.3(1) Special Investigation $f < 8$ Hz

7	Acceleration Resonance Walking	$a < 0,1 \text{ m/s}^2$	better performance	7.3.3.(1)
8	Acceleration Resonance Walking	$a < 0,35 \text{ bis } 0,7 \text{ m/s}^2$	poorer performance	7.3.3 (1)

Compare:

Kreuzinger, H.; Blaß, H.J.; Ehlbeck, J.; Steck, G.:

Erläuterungen zu DIN 1052:2004-08 –

Entwurf, Berechnung und Bemessung von Holzbauwerken.

Hrsg.: DGfH, Bruderverlag, Albert Bruder GmbH, Karlsruhe

# Damping ratio

$$D, \zeta = R / (2 M \omega)$$

$$\delta, \Lambda = \text{logarithmic Damping ratio, } 2 \pi D, 2 \pi \zeta$$

$$\Lambda = \ln \frac{w_i}{w_{i+1}}$$

$$D \cong \frac{\Lambda}{2\pi}$$

Some values for D:

steel 0,005

concret 0,008

timber 0,010 bis 0,02

Wenn keine genaueren Informationen vorliegen, ist das Dämpfungsmaß  $\zeta$  (logarithmisches Dekrement geteilt durch  $2\pi$ ) ein Wert von 0,01 anzunehmen. Weitere Richtwerte sind:

-Holztragwerke ohne mech. Verbindungen	0,010
-Holztragwerke mit mech. Verbindungen	0,015
-Holzdecken ohne schwimmenden Estrich	0,010
-Decken aus Brettschichtholz mit schwimmenden Estrich	0,020
-Holzbalkendecken aus Brettstapeldecken mit einem schwimmenden Estrich	0,030



EN 1995-1-1:2004 (E)

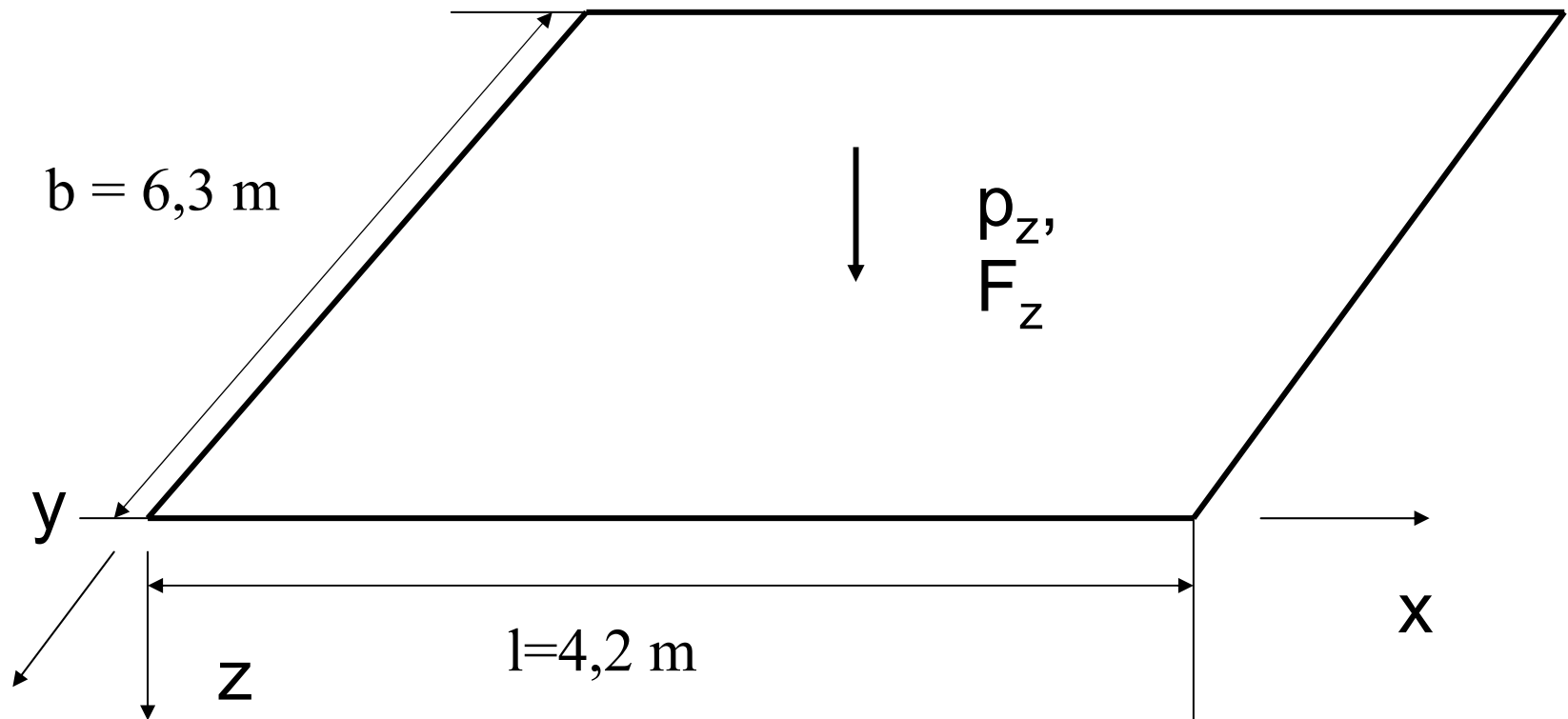
Eurocode 5

Design of timber structures

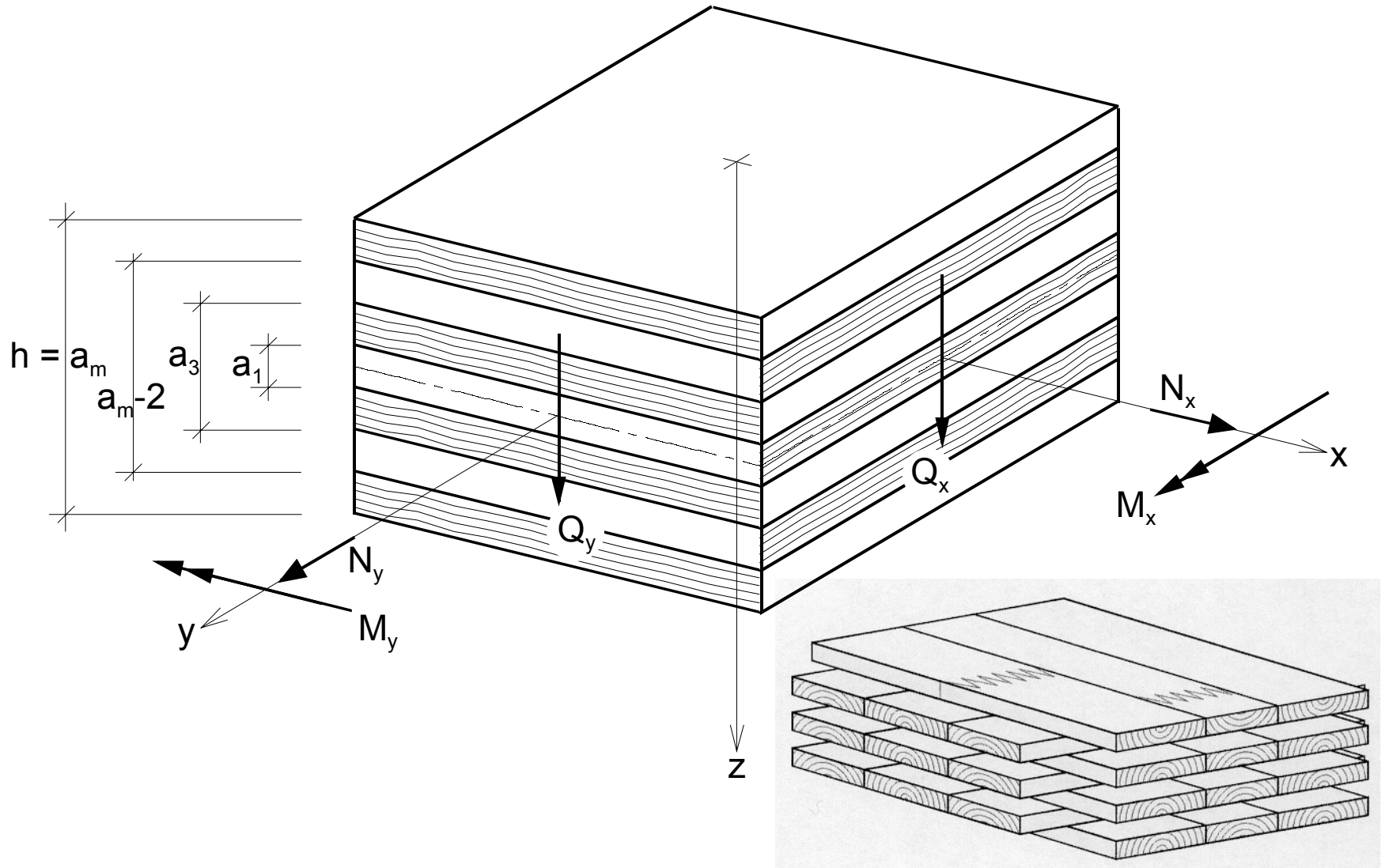
Part 1-1: General –  
Common rules and rules for buildings

Schweizer Norm  
Swiss code  
SIA 265

# Example - System



# BSP

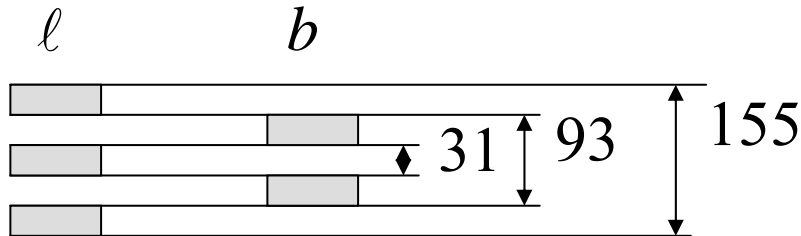


Self weight	3,23 kN/m <sup>2</sup>
Traffic load	2,00 kN/m <sup>2</sup>
Quasi ständige Kombination nach DIN 1055-100, Gl 24	

$$q_s = g + \Psi_2 \cdot p = 3,23 + 0,3 \cdot 2,0 = 3,83 \text{ kN} / \text{m}^2$$

# Bending Stiffness

Decke: BSP 155 mm 5 Lagen je 31 mm



Biegesteifigkeiten:

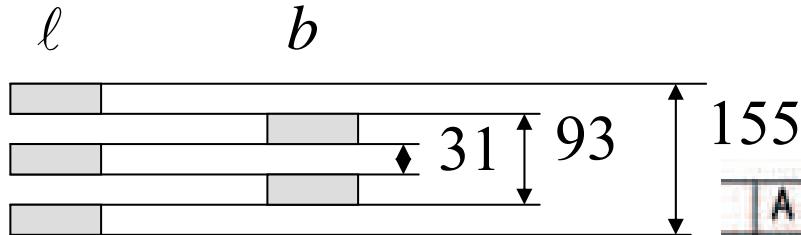
$$EI_{\ell} = 11000 \cdot \frac{1}{12} \cdot (0,155^3 - 0,093^3 + 0,031^3) = 2,70 \text{ MNm}^2$$

$$EI_b = 11000 \cdot \frac{1}{12} \cdot (0,093^3 - 0,031^3) = 0,71 \text{ MNm}^2$$

$$EI_{\ell+b} = 11000 \cdot \frac{1}{12} \cdot 0,155^3 = 3,41 \text{ MNm}^2$$



# Decke: BSP 155 mm 5 Lagen je 31 mm



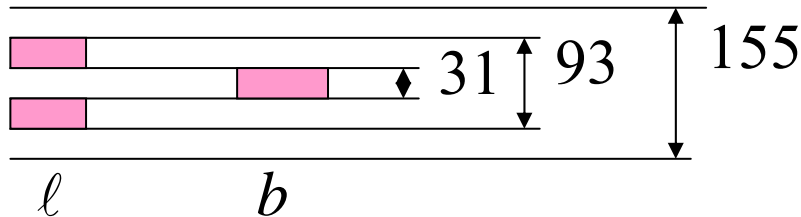
Biegesteifigkeiten:

A netto	I netto	W netto	i netto	A <sub>q</sub>
A netto	I netto	W netto	i netto	A <sub>q</sub>
A net	I net	W net	i net	A <sub>q</sub>
A netto	I netto	W netto	i netto	A <sub>q</sub>
[ cm <sup>2</sup> ]	[ cm <sup>3</sup> ]	[ cm <sup>4</sup> ]	[ cm ]	[ cm <sup>2</sup> ]
500	3.385	903	2,60	116
500	4.180	1.032	2,89	123
620	6.455	1.388	3,23	143
750	12.891	2.063	4,15	229
810	14.554	2.222	4,24	251
810	17.914	2.505	4,70	271
930	24.578	3.171	5,14	284

$$I_{\ell} = \frac{1}{12} \cdot (0,155^3 - 0,093^3 + 0,031^3) = 2,46 \cdot 10^{-4} m^4 = 24600 cm^4$$

# Shear Stiffness

Decke: BSP 155 mm 5 Lagen je 31 mm



$$a_\ell = 155 - 31 = 124 \text{ mm}$$

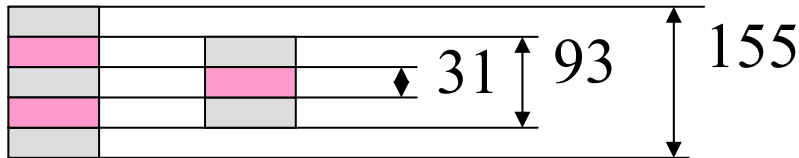
$$a_b = 93 - 31 = 62 \text{ mm}$$

$$S_\ell = \frac{a_\ell^2 \cdot G_R \cdot 1}{n \cdot d_i} = \frac{0,124^2 \cdot 75 \cdot 1}{2 \cdot 0,031} = 18,6 \text{ MN}$$

$$S_b = \frac{a_b^2 \cdot G_R \cdot 1}{n \cdot d_i} = \frac{0,062^2 \cdot 75 \cdot 1}{1 \cdot 0,031} = 9,3 \text{ MN}$$

# Effective Stiffness

Decke: BSP 155 mm 5 Lagen je 31 mm



Näherung:

$$efEI = EI \cdot \frac{1}{1 + \frac{EI \cdot \pi^2}{S \cdot l^2}}$$

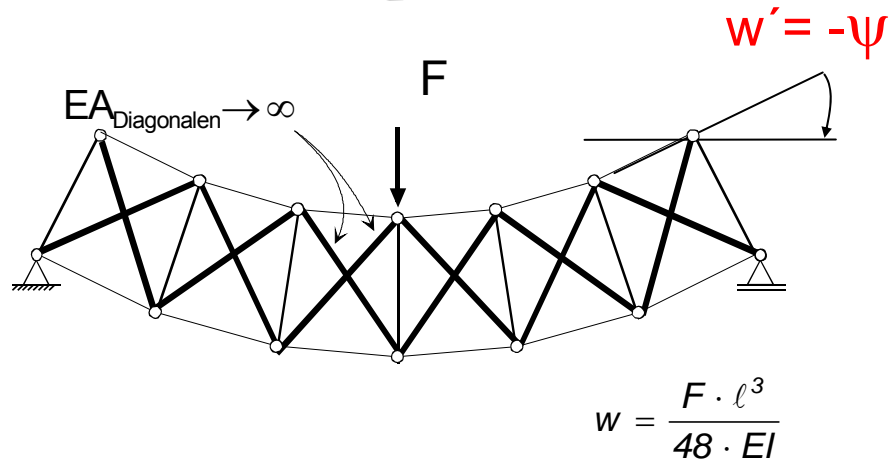
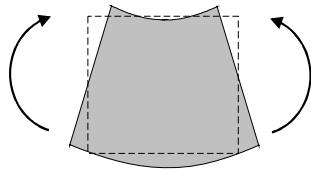
Wirksame Steifigkeiten:

$$efEI_{\ell} = 2,704 \cdot \frac{1}{1 + \frac{2,704 \cdot \pi^2}{18,6 \cdot 4,2^2}} = 2,50 \text{ MNm}^2$$

$$efEI_b = 0,71 \cdot \frac{1}{1 + \frac{0,71 \cdot \pi^2}{9,3 \cdot 6,3^2}} = 0,70 \text{ MNm}^2$$

# Beispiel – Biege- Schubverformung

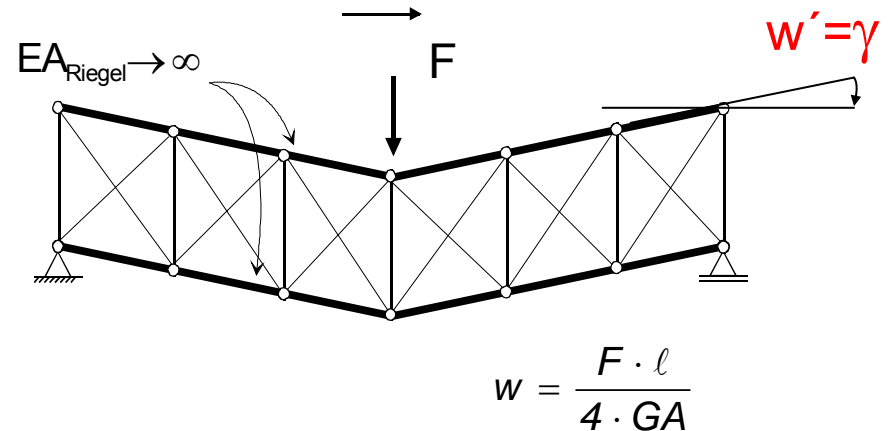
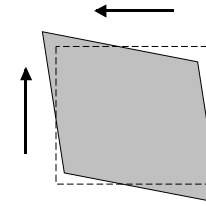
Biegung ohne Schub  
( $EI, GA \rightarrow \infty$ )



Im Fachwerkmodell:

- Elastische Gurte ( $EA$ ),
- Dehnstarre Diagonalen ( $EA \rightarrow \infty$ )

Reine Schubverformung  
( $GA, EI \rightarrow \infty$ )



Im Fachwerkmodell:

- Dehnstarre Gurte ( $EA \rightarrow \infty$ ),
- Elastische Diagonalen ( $EA$ )

# Deflection, Frequency

## Deflection, single span

$$w = \frac{5 \cdot g \cdot \ell^4}{384 \cdot efEI_\ell} = \frac{5 \cdot 3,83 \cdot 10^{-3} \cdot 4,2^4}{384 \cdot 2,50}$$
$$= 6,21 \cdot 10^{-3} \text{ m} = 6,21 \text{ mm}$$

## Frequency

$$f = \frac{5}{\sqrt{0,8 \cdot w}} = \frac{5}{\sqrt{0,8 \cdot 0,621}} = 7,09 \text{ Hz}$$

# Frequency plate

Anisotrope Platte, Trägerrost

$m^*$ : Masse (tausend t/m<sup>2</sup>)

$k$ : Zahl der Wellen in x-Richtung

$n$ : Zahl der Wellen in y-Richtung

$n_{40}$ : Zahl der Eigenfrequenzen unter 40 Hz

$$\psi_{k,n} = w_{k,n} \cdot \sin \frac{k\pi}{\ell} \cdot x \cdot \sin \frac{n\pi}{b} y$$

$$m\omega^2 = EJ_{\ell} \frac{k^4 \pi^4}{\ell^4} + EJ_b \frac{n^4 \pi^4}{b^4}; \quad f_0 = \frac{\pi}{2\ell^2} \cdot \sqrt{\frac{EJ_{\ell}}{m}}$$

$$f_{m,n} = f_0 \sqrt{k^4 + \frac{EJ_b}{EJ_{\ell}} \cdot \frac{\ell^4}{b^4} \cdot n^4}$$

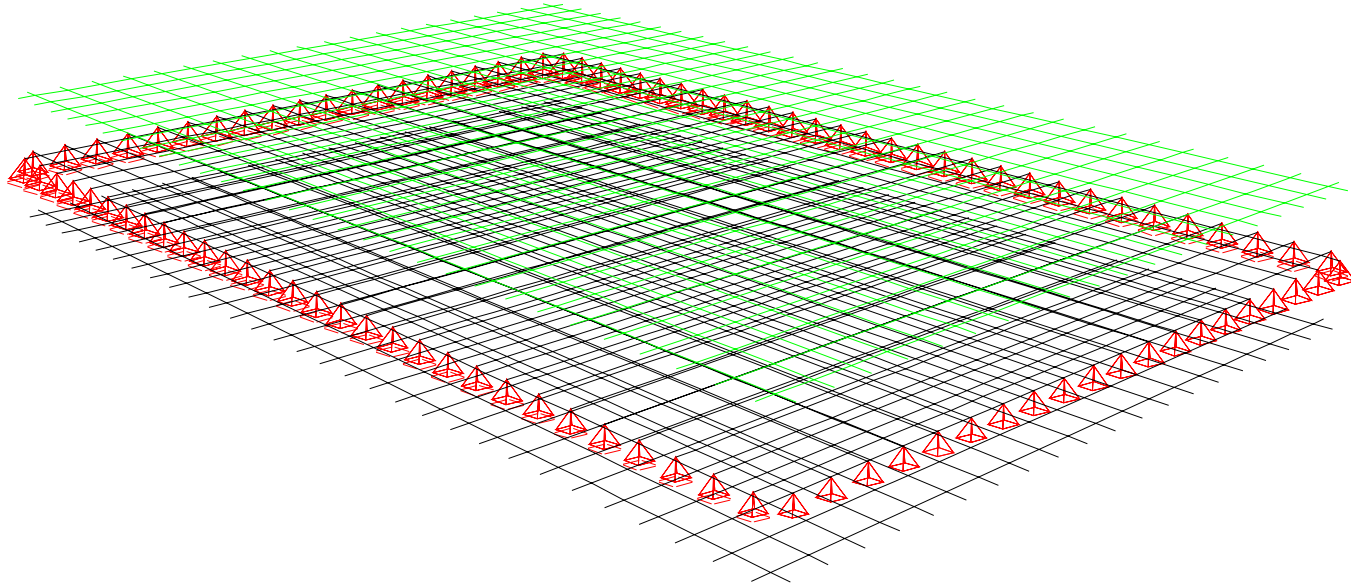
# Frequency plate

$$f_0 = \frac{\pi}{2 \cdot \ell^2} \cdot \sqrt{\frac{efEI}{m}} = \frac{\pi}{2 \cdot 4,2^2} \cdot \sqrt{\frac{2,50}{0,383 \cdot 10^{-3}}} = 7,19 \text{ Hz}$$

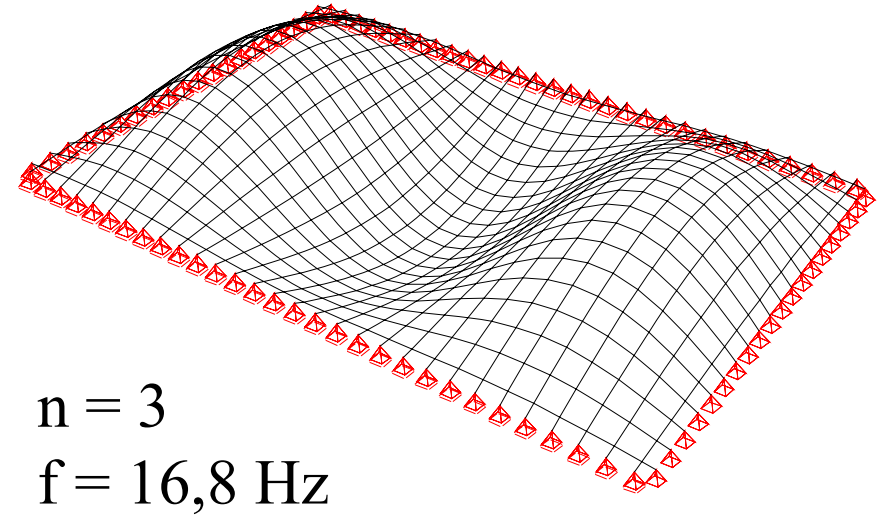
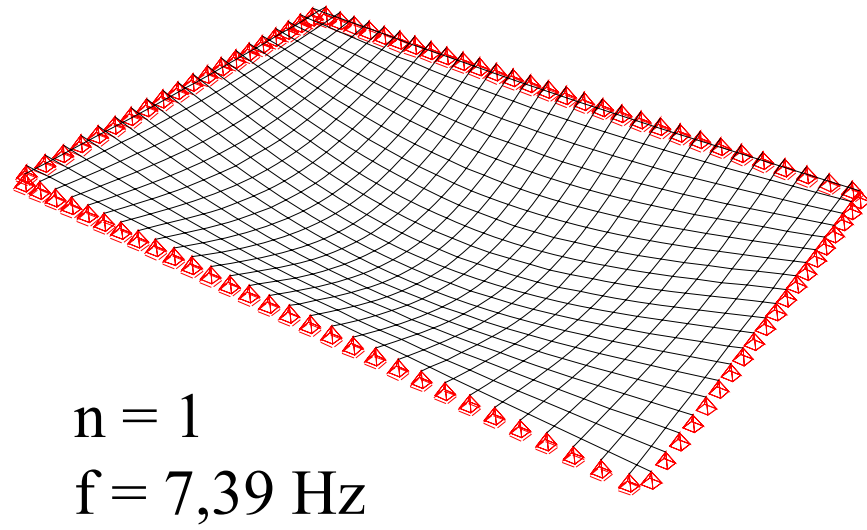
$$\sqrt{k^4 + \frac{EI_b \cdot \ell^4}{efEI \cdot b^4} \cdot n^4} = \sqrt{k^4 + \frac{0,7 \cdot 4,2^4}{2,50 \cdot 6,3^4} \cdot n^4} = \sqrt{k^4 + 0,055 \cdot n^4}$$

k Wellen in Tragrichtung	n Wellen senkrecht zur Tragrichtung						
	1	2	3	4	5		
1	7,39	8,88	16,8	27,9	42,8		
2	28,8	29,5	32,5	39,4			
3	66,5						

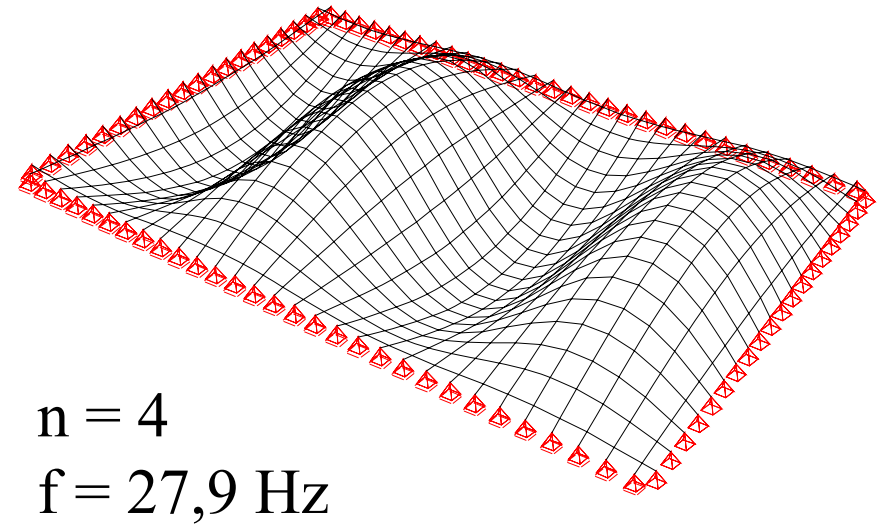
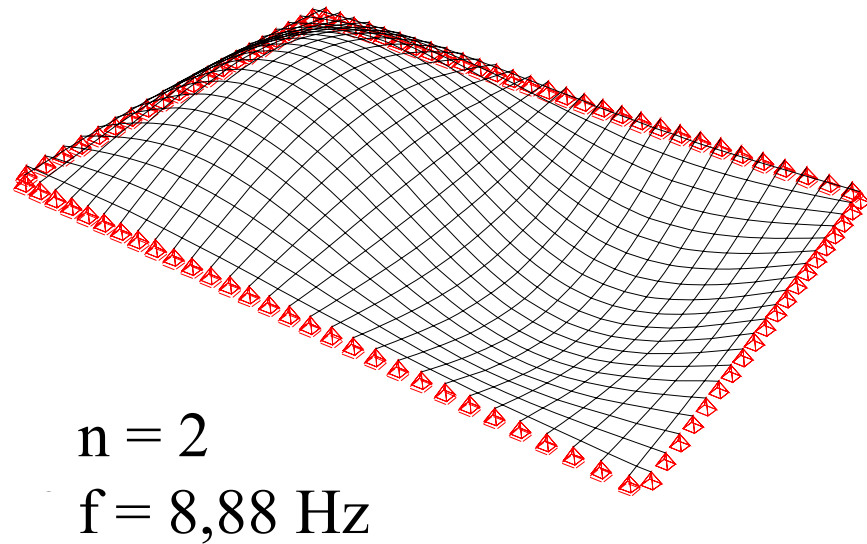
# Frequencies



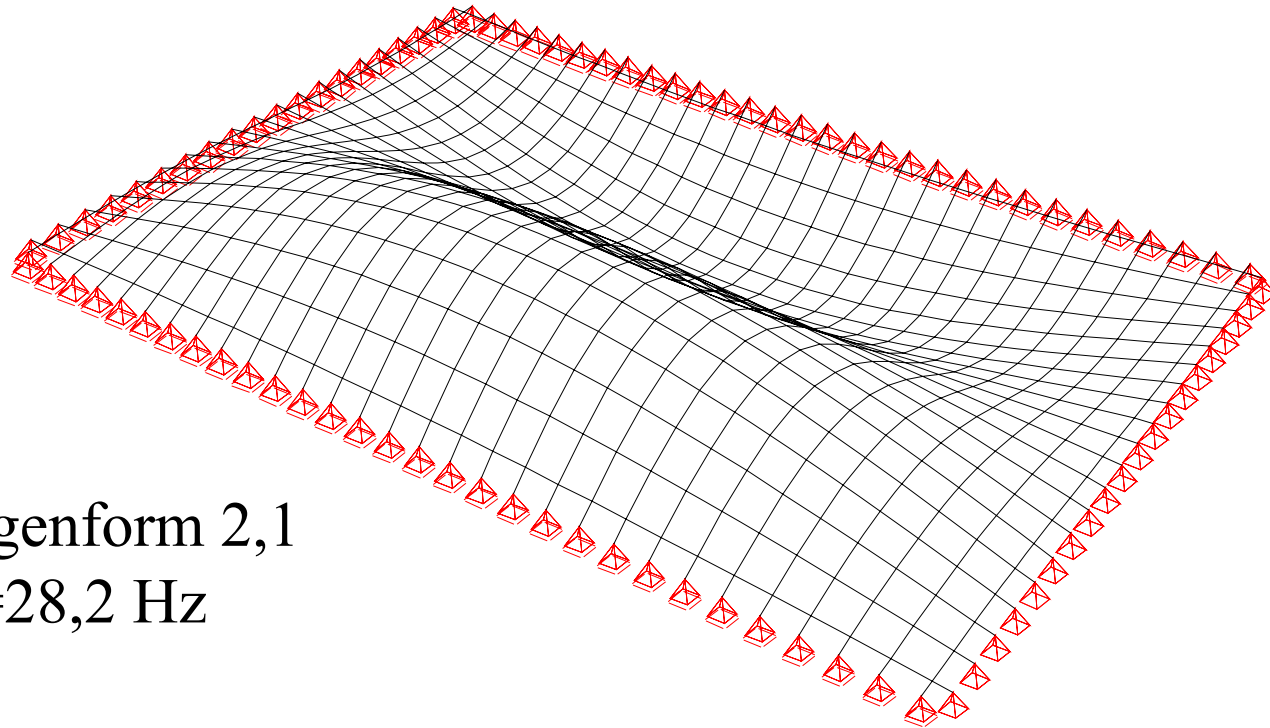




$k=1$



$k=2$



Eigenform 2,1  
 $f = 28,2 \text{ Hz}$

## deflection

$k_{def} = 0,8$ , EDIN 1052, Tble 3.2

$$efEI(k_{def}) = \frac{efEI}{1 + k_{def}}$$

Quasi ständige Bemessungssituation nach Gleichung (42)  
DIN 1052

1	$(1 + k_{def}) \cdot W_{G,inst}$ 1,8 · 5,25 = 9,4 mm	$\psi_2 \cdot (1 + k_{def}) \cdot W_{Q,inst}$ 0,3 · (1+0,8) · 3,25 = 1,8 mm	11,2 mm	0,0112/ 4,2 = 1/375
---	--	---	---------	---------------------------

1	deflection w	$w < l/X$	Enough rigidity Small deflection	7.2
---	-----------------	-----------	-------------------------------------	-----

## Deflection single load

$$\underline{F = 1 \text{ kN}}$$

$$w_F = \frac{F \cdot \ell^3}{48 \cdot EI_\ell \cdot b_F} = \frac{1 \cdot 10^{-3} \cdot 4,2^3}{48 \cdot 2,50 \cdot 2,78} = 0,22 \cdot 10^{-3} \text{ m} = 0,22 \text{ mm}$$

$$b_F = \frac{\ell}{1,1} \cdot \sqrt[4]{\frac{EI_b}{EI_\ell}} = \frac{4,2}{1,1} \cdot \sqrt[4]{\frac{0,7}{2,50}} = 2,78 \text{ m}$$

0,22 < 0,5 bis 4 mm

3	Durchbiegung Einzellast F=1kN	u <0,5 bis 4 mm	Querverteilung geringe Verformung	DIN ENV 1995-1-1 7.3.3 SIA 265
---	-------------------------------------	--------------------	---	---

# Frequency

$$f_{1,1} = 7,19 \text{ Hz} < 8 \text{ Hz} !!$$

4	frequency	$f > 8 \text{ Hz}$	No resonance	7.3.3(2)
	$w_{G,inst}$ quasi ständig, $g + \psi_2 p$	( $w < 0,5 \text{ cm}$ )		7.3.3(1)
		$f < 8 \text{ Hz}$		
		( $w > 0,5 \text{ cm}$ )		

## Impuls I=1 Ns

$$v = \frac{4 \cdot (0,4 + 0,6 \cdot n_{40})}{m \cdot b \cdot \ell + 200} = \frac{4 \cdot (0,4 + 0,6 \cdot 5)}{383 \cdot 4,2 \cdot 6,3 + 200} = 0,00132 \frac{m}{s}$$

„better performance“

$$b^{(f \cdot \xi - 1)} = 100^{(7,19 \cdot 0,01 - 1)} = 0,014 \frac{m}{s}$$

5	Impuls I=1 Ns (up to 40 Hz)	$v < b^{(f \cdot \xi - 1)}$  $50 < b < 150$ $\xi = 0,01$	Velocity	7.3.3(2) SIA 265
---	-----------------------------------	---	----------	---------------------

heeldrop  $I=55 \text{ Ns}$ 

$$v = \frac{0,6}{m^{0,5} \cdot EI_{\ell}^{0,25} \cdot EI_b^{0,25}} =$$

$$\frac{0,6}{383^{0,5} \cdot 2,50^{0,25} \cdot 0,70^{0,25}} = 0,027 \frac{m}{s}$$

6	velocity heeldrop $I=55\text{Ns}$ , $t_i = 0,05\text{s}$	$v < 6 b_1^{(f_1 \zeta - 1)}$	velocity	Mohr /bmh/
---	---	-------------------------------	----------	------------

$$6 \cdot 0,014 = 0,084 \text{ m/s} > 0,027$$



7.3.3 (1)  $f < 8$  Hz: Special investigationDecke:  $f = 7,19$  Hz

Gehen: Resonanz

$$f_s = 7,19 / 3 = 2,4 \text{ Hz}$$

$$\frac{\alpha_3 \cdot F_0}{M} \cdot V(3f_s / f_{1,1}) = \frac{0,1 \cdot 700}{2520} \cdot 33,3 = 0,925 \frac{m}{s^2}$$

$$0,4 \cdot 0,93 = 0,37 \text{ m/s}^2 > 0,1 \text{ m/s}^2$$

7	Beschleunigung Resonanz- untersuchung	$a < 0,1 \text{ m/s}^2$	Beschleunigung Wohlbefinden	
---	---	-------------------------	--------------------------------	--

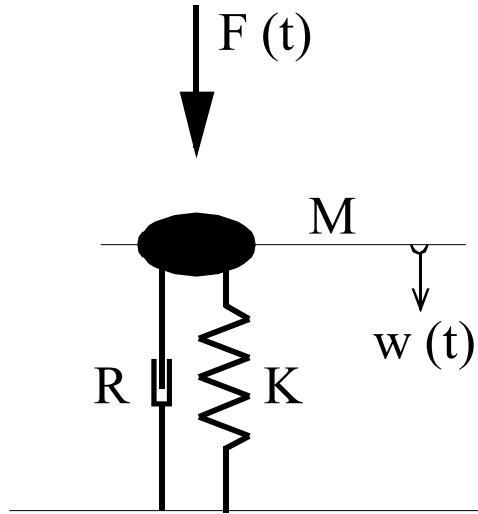
## Gehen: Resonanz

$$f_s = 7,19 / 3 = 2,4 \text{ Hz}$$

$$\frac{\alpha_3 \cdot F_0}{M} \cdot V(3f_s / f_{1,1}) = \frac{0,1 \cdot 700}{2520} \cdot 33,3 = 0,925 \frac{m}{s^2}$$

$$0,4 \cdot 0,93 = 0,37 \text{ m/s}^2 < 0,7 \text{ m/s}^2$$

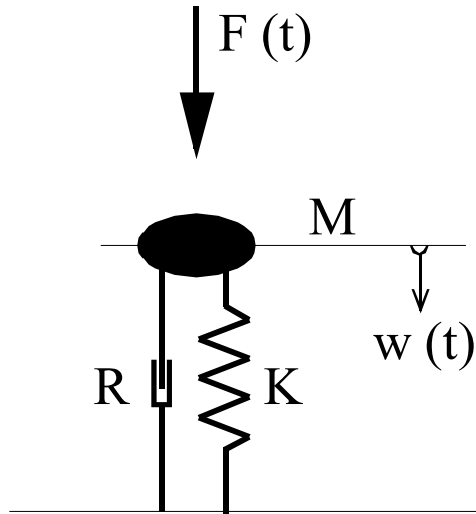
8	Beschleunigung Resonanz- untersuchung	$a < 0,35$ bis $0,7$ $m/s^2$	Beschleunigung Spürbar, nicht störend	DIN1052, 9.3(3) besondere Unter- suchungen
---	---	---------------------------------	---	--



Deformation

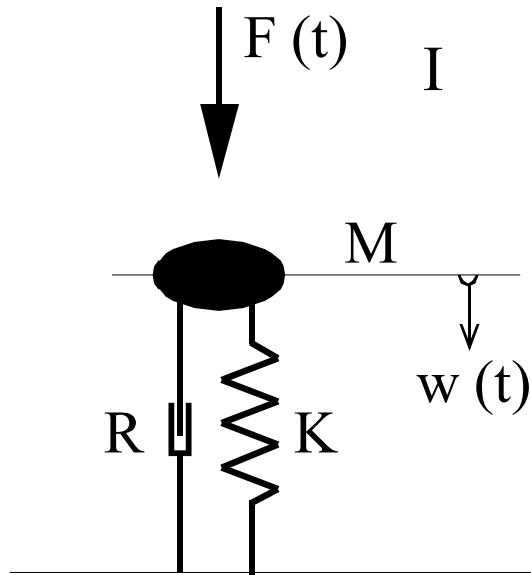
## Summary

$$w \left( \frac{1}{Rigidity} \right)$$



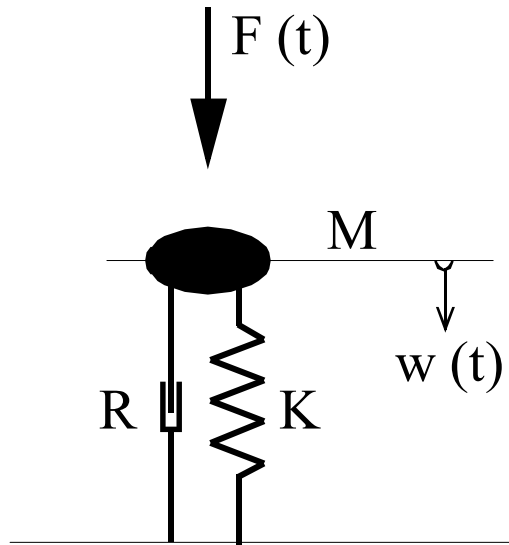
Frequency

$$f = f \left( \sqrt{\frac{\textit{rigidity}}{\textit{mass}}} \right)$$



Velocity

$$v \left( \frac{1}{mass} \right)$$



Acceleration

$$a \left( \begin{array}{c} \frac{1}{\text{mass}} \quad \frac{1}{\text{damping}} \end{array} \right)$$

Wanted:

Timber floors

great stiffness

most

high mass

no

high frequency

not always

high damping

yes

Thank you very much  
for your attention!

Vielen Dank für's Zuhören !